## LMI Methods in Optimal and Robust Control

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Lecture 11: Relationship between  $H_2$ , LQG and LGR and LMIs for state and output feedback  $H_2$  synthesis

To solve the  $H_\infty\mbox{-}{\rm optimal}$  output-feedback problem, we solve

$$\begin{array}{l} \min_{\gamma, X_1, Y_1, A_n, B_n, C_n, D_n} \gamma \quad \text{such that} \\ \begin{bmatrix} X_1 & I \\ I & Y_1 \end{bmatrix} > 0 \\ \begin{bmatrix} AY_1 + Y_1 A^T + B_2 C_n + C_n^T B_2^T & *^T & *^T & *^T \\ A^T + A_n + [B_2 D_n C_2]^T & X_1 A + A^T X_1 + B_n C_2 + C_2^T B_n^T & *^T & *^T \\ [B_1 + B_2 D_n D_{21}]^T & [XB_1 + B_n D_{21}]^T & -\gamma I \\ C_1 Y_1 + D_{12} C_n & C_1 + D_{12} D_n C_2 & D_{11} + D_{12} D_n D_{21} - \gamma I \end{bmatrix} < 0$$

## Conclusion

Then, we construct our controller using

$$D_{K} = (I + D_{K2}D_{22})^{-1}D_{K2}$$
  

$$B_{K} = B_{K2}(I - D_{22}D_{K})$$
  

$$C_{K} = (I - D_{K}D_{22})C_{K2}$$
  

$$A_{K} = A_{K2} - B_{K}(I - D_{22}D_{K})^{-1}D_{22}C_{K}.$$

where

$$\begin{bmatrix} A_{K2} & B_{K2} \\ \hline C_{K2} & D_{K2} \end{bmatrix} = \begin{bmatrix} X_2 & X_1 B_2 \\ 0 & I \end{bmatrix}^{-1} \begin{bmatrix} A_n & B_n \\ C_n & D_n \end{bmatrix} - \begin{bmatrix} X_1 A Y_1 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} Y_2^T & 0 \\ C_2 Y_1 & I \end{bmatrix}^{-1}$$

and where  $X_2$  and  $Y_2$  are any matrices which satisfy  $X_2Y_2^T = I - X_1Y_1$ .

- e.g. Let  $Y_2 = I$  and  $X_2 = I X_1 Y_1$ .
- The optimal controller is NOT uniquely defined.
- Don't forget to check invertibility of  $I D_{22}D_K$

The  $H_\infty$ -optimal controller is a dynamic system.

• Transfer Function 
$$\hat{K}(s) = \begin{bmatrix} A_K & B_K \\ \hline C_K & D_K \end{bmatrix}$$

Minimizes the effect of external input (w) on external output (z).

$$||z||_{L_2} \le ||\underline{\mathsf{S}}(P, K)||_{H_{\infty}} ||w||_{L_2}$$

• Minimum Energy Gain

Motivation

 $H_2$ -optimal control minimizes the  $H_2$ -norm of the transfer function.

• The *H*<sub>2</sub>-norm has no direct interpretation.

$$\|G\|_{H_2}^2 = \frac{1}{2\pi} \int_{-\infty}^{\infty} \mathrm{Trace}(\hat{G}(\imath\omega)^* \hat{G}(\imath\omega)) d\omega$$

Motivation: Assume external input, w, is Gaussian noise with power spectral density  $\hat{S}_w.$  Then, the variance is given by

$$E[w(t)^2] = \frac{1}{2\pi} \int_{-\infty}^{\infty} \operatorname{Trace}(\hat{S}_w(\imath\omega)) d\omega$$

#### Theorem 1.

For an LTI system P, if w is noise with spectral density  $\hat{S}_w(\iota\omega)$  and z = Pw, then z is noise with density

$$\hat{S}_z(\iota\omega) = \hat{P}(\iota\omega)\hat{S}(\iota\omega)\hat{P}(\iota\omega)^*$$

Then the output z = Gw has signal variance (Power)

$$\begin{split} E[z(t)^2] &= \frac{1}{2\pi} \int_{-\infty}^{\infty} \operatorname{Trace}(\hat{G}(\imath\omega)^* S(\imath\omega) \hat{G}(\imath\omega)) d\omega \\ &\leq \|S\|_{H_{\infty}} \|G\|_{H_2}^2 \end{split}$$

If the input signal is white noise, then  $\hat{S}(\imath\omega)=I$  and

$$E[z(t)^2] = \|\hat{G}\|_{H_2}^2$$

Lecture 11

#### $-H_2$ -optimal control

 $H_{2}$ -optimal control Motivation

Then the output z = Gw has signal variance (Power)  $E[z(t)^2] = \frac{1}{2\pi} \int_{-\infty}^{-\infty} Trace(\hat{G}(w)^* S(w)\hat{G}(w))d\omega$   $\leq |S||_{H_{w}} |G||_{H_{w}}^{2}$ If the input signal is white noise, then  $\hat{S}(w) = I$  and  $E[z(t)^2 = |G||_{H_{w}}^{2}$ 

Hence the  $H_2$  norm represents the power spectral density of the output of the system when the input is white noise.

- Thus  $H_2$  optimal control is optimal in a certain sense when the input is expected to be white noise.
- However, this doesn't work when the noise is colored (concentrated at certain frequencies).
- For colored noise, however, we can use prefilters to obtain optimal controllers.

Colored Noise

Now suppose the noise is colored with density  $\hat{S}_w(\imath\omega)$ . Now define  $\hat{H}$  as  $\hat{H}(\imath\omega)\hat{H}(\imath\omega)^* = \hat{S}_w(\imath\omega)$  and the filtered system

$$\hat{P}_s(s) = \begin{bmatrix} \hat{P}_{11}(s)\hat{H}(s) & \hat{P}_{12}(s)\\ \hat{P}_{21}(s)\hat{H}(s) & \hat{P}_{22}(s) \end{bmatrix}$$

Now, applying feedback to the filtered plant, we get

$$\underline{S}(P_s, K)(s) = P_{11}H + P_{12}(I - KP_{22})^{-1}KP_{21}H = \underline{S}(P, K)H$$

Now the spectral density,  $\hat{S}_z$  of the output of the true plant using colored noise equals the output of the artificial plant under white noise. i.e.

$$\begin{split} \hat{S}_z(s) &= \underline{\mathsf{S}}(P, K)(s)\hat{S}_w(s)\underline{\mathsf{S}}(P, K)(s)^* \\ &= \underline{\mathsf{S}}(P, K)(s)\hat{H}(s)\hat{H}(s)^*\underline{\mathsf{S}}(P, K)(s)^* = \hat{S}(P_s, K)(s)\hat{S}(P_s, K)(s)^* \end{split}$$

Thus if K minimizes the  $H_2$ -norm of the filtered plant  $(\|\hat{S}(P_s, K)\|_{H_2}^2)$ , it will minimize the variance of the true plant under the influence of colored noise with density  $\hat{S}_w$ .

In this case, the response of the prefiltered system to white noise is the same as the unfiltered system response to colored noise.

Alternatively, we can write

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$$\min_{K} \|S(P,K)w\|_{w=colored} = \min_{K} \|S(P,K)Hu\|_{u=white} = \min_{K} \|S(P,K)H\|_{H_2}$$

#### Theorem 2.

Suppose  $\hat{P}(s) = C(sI - A)^{-1}B$ . Then the following are equivalent.

- 1. A is Hurwitz and  $\|\hat{P}\|_{H_2} < \gamma$ .
- **2**. There exists some X > 0 such that

$$\operatorname{trace} B^T X B < \gamma^2$$
$$A^T X + X A + C^T C < 0$$

Recall that the Controllability Grammian is a solution!

- Recall how the proof works.
- But this time use the observability grammian.

#### Proof.

Suppose A is Hurwitz and  $\|\hat{P}\|_{H_2} < \gamma.$  Then the Observability Grammian is defined as

$$X_o = \int_0^\infty e^{A^T t} C^T C e^{At} dt$$

Now recall the Laplace transform

$$(\Lambda e^{At})(s) = \int_0^\infty e^{At} e^{-ts} dt$$
$$= \int_0^\infty e^{-(sI-A)t} dt$$
$$= -(sI-A)^{-1} e^{-(sI-A)t} dt \Big|_{t=0}^{t=-\infty}$$
$$= (sI-A)^{-1}$$

Hence  $(\Lambda C e^{At} B)(s) = C(sI - A)^{-1}B.$ 

#### Proof.

$$\begin{split} \left(\Lambda Ce^{At}B\right)(s) &= C(sI-A)^{-1}B \text{ implies} \\ \|\hat{P}\|_{H_2}^2 &= \|C(sI-A)^{-1}B\|_{H_2}^2 \\ &= \frac{1}{2\pi}\int_0^\infty \operatorname{Trace}((C(\omega I-A)^{-1}B)^*(C(\iota\omega I-A)^{-1}B))d\omega \\ &= \operatorname{Trace}\int_{-\infty}^\infty B^T e^{A^T t} C^T Ce^{At}Bdt \\ &= \operatorname{Trace}B^T X_o B \end{split}$$

Thus  $X_o \ge 0$  and  $\operatorname{Trace} B^T X_o B = \|\hat{P}\|_{H_2}^2 < \gamma^2$ .

The rest of the proof we can skip.

Lets consider the full-state feedback problem

$$\hat{G}(s) = \begin{bmatrix} A & B_1 & B_2 \\ \hline C_1 & 0 & D_{12} \\ I & 0 & 0 \end{bmatrix}$$

- $D_{12}$  is the weight on control effort.
- $D_{11} = 0$  is a feed-through term and must be 0.
- $C_2 = I$  as this is state-feedback.

$$\hat{K}(s) = \begin{bmatrix} 0 & 0\\ \hline 0 & K \end{bmatrix}$$

Full-State Feedback

#### Theorem 3.

The following are equivalent.

1.  $||S(K, P)||_{H_2} < \gamma$ . 2.  $K = ZX^{-1}$  for some Z and X > 0 where  $\begin{bmatrix} A & B_2 \end{bmatrix} \begin{bmatrix} X \\ Z \end{bmatrix} + \begin{bmatrix} X & Z^T \end{bmatrix} \begin{bmatrix} A^T \\ B^T \end{bmatrix} + B_1 B_1^T < 0$ 

*Trace* 
$$[C_1X + D_{12}Z] X^{-1} [C_1X + D_{12}Z] < \gamma^2$$

However, this is nonlinear, so we need to reformulate using the Schur Complement.

Applying the Schur Complement gives the alternative formulation convenient for control.

#### Theorem 4.

Suppose  $\hat{P}(s) = C(sI - A)^{-1}B$ . Then the following are equivalent.

- 1. A is Hurwitz and  $\|\hat{P}\|_{H_2} < \gamma$ .
- 2. There exists some X, W > 0 such that

$$\begin{bmatrix} A^T X + X A & X B \\ B^T X & -\gamma I \end{bmatrix} < 0, \qquad \begin{bmatrix} X & C^T \\ C & W \end{bmatrix} > 0, \qquad \textit{TraceW} < \gamma^2$$

Full-State Feedback

#### Theorem 5.

The following are equivalent.

1.  $||S(K, P)||_{H_2} < \gamma$ . 2.  $K = ZX^{-1}$  for some Z and X > 0 where  $\begin{bmatrix} A & B_2 \end{bmatrix} \begin{bmatrix} X \\ Z \end{bmatrix} + \begin{bmatrix} X & Z^T \end{bmatrix} \begin{bmatrix} A^T \\ B_2^T \end{bmatrix} + B_1 B_1^T < 0$   $\begin{bmatrix} X & (C_1X + D_{12}Z)^T \\ C_1X + D_{12}Z & W \end{bmatrix} > 0$ 

$$\mathit{TraceW} < \gamma^2$$

Thus we can solve the  $H_2$ -optimal static full-state feedback problem.

The LQR Problem:

- Full-State Feedback
- Choose K to minimize the cost function

$$\int_0^\infty x(t)^T Q x(t) + u(t)^T R u(t) dt$$

subject to dynamic constraints

$$\dot{x}(t) = Ax(t) + Bu(t)$$
$$u(t) = Kx(t), \qquad x(0) = x_0$$

Trying to minimize the effect of  $x_0$  on a weighted- $L_2$ -norm of the regulated output.

Relationship to LQR

To solve the LQR problem using  $H_2$  optimal state-feedback control, let

• 
$$C_1 = \begin{bmatrix} Q^{\frac{1}{2}} \\ 0 \end{bmatrix}$$
,  
•  $D_{12} = \begin{bmatrix} 0 \\ R^{\frac{1}{2}} \end{bmatrix}$  and  $D_{11} = 0$ ,  
•  $B_2 = B$  and  $B_1 = I$ .

So that

$$\underline{S}(P,K) = \begin{bmatrix} A + B_2 K & B_1 \\ \hline C_1 + D_{12} K & D_{11} \end{bmatrix} = \begin{bmatrix} A + B K & I \\ \hline Q^{\frac{1}{2}} \\ R^{\frac{1}{2}} K & 0 \end{bmatrix}$$

And solve the  $H_2$  full-state feedback problem. Then if

$$\dot{x}(t) = A_{CL}x(t) = (A + BK)x(t) = Ax(t) + Bu(t)$$
  
 $u(t) = Kx(t), \qquad x(0) = x_0$ 

Then  $x(t) = e^{A_{CL}t}x_0$ .

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Translating to the input-output formulation, recall we apply the problem setup to

$$P = \begin{bmatrix} A & B_1 & B_2 \\ \hline C_1 \\ I \end{bmatrix} \begin{bmatrix} D_{11} & D_{12} \\ 0 & 0 \end{bmatrix}$$
$$K = \begin{bmatrix} 0 & 0 \\ \hline 0 & K \end{bmatrix}$$

# $H_2$ -optimal control Relationship to LQR

Ignoring the regulated outputs for now, we have

$$\dot{x}(t) = A_{CL}x(t) = (A + BK)x(t) = Ax(t) + Bu(t)$$
  
 $u(t) = Kx(t), \qquad x(0) = x_0$ 

then  $x(t)=e^{A_{CL}t}x_0\text{, }u(t)=Ke^{A_{CL}t}x_0$  and

$$\begin{split} &\int_{0}^{\infty} x(t)^{T}Qx(t) + u(t)^{T}Ru(t)dt = \int_{0}^{\infty} x_{0}^{T}e^{A_{CL}^{T}t}(Q + K^{T}RK)e^{A_{CL}t}x_{0}dt \\ &= \mathsf{Trace}\int_{0}^{\infty} x_{0}^{T}e^{A_{CL}^{T}t} \begin{bmatrix} Q^{\frac{1}{2}} \\ R^{\frac{1}{2}}K \end{bmatrix}^{T} \begin{bmatrix} Q^{\frac{1}{2}} \\ R^{\frac{1}{2}}K \end{bmatrix} e^{A_{CL}t}x_{0}dt \\ &= \|x_{0}\|^{2}\mathsf{Trace}\int_{0}^{\infty} B_{1}^{T}e^{A_{CL}^{T}t}(C_{1} + D_{12}K)^{T}(C_{1} + D_{12}K)e^{A_{CL}t}B_{1}dt \\ &= \|x_{0}\|^{2}B_{1}^{T}X_{0}B_{1} = \|x_{0}\|^{2}\|S(P,K)\|_{H_{2}}^{2} \end{split}$$

Thus LQR reduces to a special case of  $H_2$  static state-feedback.

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#### Recall that

• 
$$C_1 = \begin{bmatrix} Q^{\frac{1}{2}} \\ 0 \end{bmatrix}$$
,  
•  $D_{12} = \begin{bmatrix} 0 \\ R^{\frac{1}{2}} \end{bmatrix}$  and  $D_{11} = 0$ ,

• 
$$B_2 = B$$
 and  $B_1 = I$ .

So that

$$\underline{\mathbf{S}}(P,K) = \begin{bmatrix} A + B_2 K & B_1 \\ \hline C_1 + D_{12} K & D_{11} \end{bmatrix} = \begin{bmatrix} A + B K & I \\ \hline Q^{\frac{1}{2}} \\ R^{\frac{1}{2}} K & 0 \end{bmatrix}$$

H2-optimal control Relationship to LQR

Ignoring the regulated outputs for now, we have

$$\begin{split} \dot{x}(t) &= A_{CL} x(t) = (A + BK) x(t) = A x(t) + B u(t) \\ u(t) &= K x(t), \qquad x(0) = x_0 \end{split}$$

then  $\boldsymbol{x}(t)=e^{AC\,i\,t}\boldsymbol{x}_0,\,\boldsymbol{u}(t)=Ke^{AC\,i\,t}\boldsymbol{x}_0$  and

$$\begin{split} &\int_0^\infty x(t)^TQ(t)+u(t)^TR(t)(dt=\int_0^\infty x_h^Te^{A_h^Tt}(t+A^TRK)e^{A_h^Tt}x_hdt)\\ &=\mathrm{Tracs}\int_0^\infty x_h^Te^{A_h^Tt}\left[\frac{dt}{dt}\right]^T\left[\frac{dt}{dt}\right]^{A_h^Tt}x_hdt\\ &=\mathrm{Tracs}\int_0^\infty x_h^Te^{A_h^Tt}\left[\frac{dt}{dt}\right]^T\left[\frac{dt}{dt}\right]^{A_h^Tt}x_hdt\\ &=\mathrm{Trac}\int_0^\infty R_h^Te^{A_h^Tt}(t+D_{2h}K)^T(t_h^T+D_{2h}K)e^{A_{2h}}B_hdt\\ &=\|x\|^2\mathrm{Tracs}\int_0^\infty R_h^Te^{A_h^Tt}(t+K)R_hdt\\ &=\|x\|^2\mathrm{Trac} R_hdt\\ &=\mathrm{Trac}(R^TA_hdt)=\mathrm{Trac}(R^TA_h)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)R^T(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_hdt)R^T(R^TA_hdt)R^T(R^TA_hdt)R^T(R^TA_hdt)\\ &=\mathrm{Trac}(R^TA_hdt)R^T(R^TA_$$

## $H_2$ -optimal output feedback control

#### Theorem 6 (Scherer, Gahinet).

The following are equivalent. • There exists a  $\hat{K} = \begin{bmatrix} A_K & B_K \\ \hline C_K & D_K \end{bmatrix}$  such that  $\|S(K, P)\|_{H_2} < \gamma$ . • There exist  $X_1, Y_1, Z, A_n, B_n, C_n, D_n$  such that  $\begin{bmatrix} AY_1 + Y_1A^T + B_2C_n + C_n^TB_2^T & *^T & *^T \\ A^T + A_n + [B_2D_nC_2]^T & X_1A + A^TX_1 + B_nC_2 + C_2^TB_n^T & *^T \\ [B_1 + B_2D_nD_{21}]^T & [X_1B_1 + B_nD_{21}]^T & -I \end{bmatrix} < 0,$  $\begin{vmatrix} Y_1 & I & *^T \\ I & X_1 & *^T \\ C_1Y_1 + D_{12}C_n & C_1 + D_{12}D_nC_2 & Z \end{vmatrix} > 0,$  $D_{11} + D_{12}D_n D_{21} = 0,$   $\operatorname{trace}(Z) < \gamma^2$ 

## $H_2$ -optimal output feedback control

As before, the controller can be recovered as

$$\begin{bmatrix} A_{K2} & B_{K2} \\ \hline C_{K2} & D_{K2} \end{bmatrix} = \begin{bmatrix} X_2 & X_1 B_2 \\ 0 & I \end{bmatrix}^{-1} \begin{bmatrix} A_n & B_n \\ C_n & D_n \end{bmatrix} - \begin{bmatrix} X_1 A Y_1 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} Y_2^T & 0 \\ C_2 Y_1 & I \end{bmatrix}^{-1}$$

for any full-rank  $X_2$  and  $Y_2$  such that

$$\begin{bmatrix} X_1 & X_2 \\ X_2^T & X_3 \end{bmatrix} = \begin{bmatrix} Y_1 & Y_2 \\ Y_2^T & Y_3 \end{bmatrix}^{-1}$$

To find the actual controller, we use the identities:

$$D_{K} = (I + D_{K2}D_{22})^{-1}D_{K2}$$
  

$$B_{K} = B_{K2}(I - D_{22}D_{K})$$
  

$$C_{K} = (I - D_{K}D_{22})C_{K2}$$
  

$$A_{K} = A_{K2} - B_{K}(I - D_{22}D_{K})^{-1}D_{22}C_{K}$$

# An LMI for Mixed $H_2$ - $H_\infty$ optimal output feedback control

#### Theorem 7.

The following are equivalent.

• There exists a  $K = \begin{bmatrix} A_K & B_K \\ \hline C_K & D_K \end{bmatrix}$  such that  $\|S(K, P)\|_{H_2} < \gamma_1$  and  $\|S(K,P)\|_{H_{\infty}} < \gamma_2.$ • There exist  $X_1, Y_1, Z, A_n, B_n, C_n, D_n$  such that  $\begin{bmatrix} AY_1 + Y_1A^T + B_2C_n + C_n^TB_2^T & *^T & *^T \\ A^T + A_n + [B_2D_nC_2]^T & X_1A + A^TX_1 + B_nC_2 + C_2^TB_n^T & *^T \\ [B_1 + B_2D_nD_{21}]^T & [X_1B_1 + B_nD_{21}]^T & -I \end{bmatrix} < 0,$  $\begin{bmatrix} Y_1 & I & *^T \\ I & X_1 & *^T \\ C_1 Y_1 + D_{12} C_n & C_1 + D_{12} D_n C_2 & Z \end{bmatrix} > 0,$  $D_{11} + D_{12}D_n D_{21} = 0,$  trace $(Z) < \gamma_1^2$  $\begin{bmatrix} AY_1 + Y_1A^T + B_2C_n + C_n^TB_2^T & *^T & *^T & *^T \\ A^T + A_n + [B_2D_nC_2]^T & X_1A + A^TX_1 + B_nC_2 + C_2^TB_n^T & *^T & *^T \\ [B_1 + B_2D_nD_{21}]^T & [X_1B_1 + B_nD_{21}]^T & -\gamma_2I & *^T \\ C_1Y_1 + D_{12}C_n & C_1 + D_{12}D_nC_2 & D_{11} + D_{12}D_nD_{21} - \gamma_2I \end{bmatrix} < 0$ 

## An LMI for the Kalman Filter! - Continuous Time

#### System:

$$\begin{split} \dot{x}(t) &= Ax(t) + Bu(t) + w(t) \\ y(t) &= Cx(t) + v(t) \end{split}$$

Filter:

$$\dot{\hat{x}}(t) = A\hat{x}(t) + Bu(t) + L(y(t) - \hat{y}(t))$$
$$\hat{y}(t) = C\hat{x}(t)$$

Error:

$$\dot{e}(t) = (A + LC)e(t) + w(t) + Lv(t)$$

The Kalman Filter chooses L to minimize the cost  $J=\mathbf{E}[e^Te].$   $L=\Sigma C^T V_2^{-1}$ 

where  $V_1 = \mathbf{E}[\mathbf{w}(\mathbf{t})\mathbf{w}(\mathbf{t})^{\mathbf{T}}]$  and  $V_2 = \mathbf{E}[\mathbf{v}(\mathbf{t})\mathbf{v}(\mathbf{t})^{\mathbf{T}}]$  and  $\Sigma$  satisfies  $A\Sigma + \Sigma A^T + V_1 = \Sigma C^T V_2^{-1} C\Sigma$ 

If we choose  $u(t) = K\hat{x}(t)$  where A + BK is stable,

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- A + LC is stable if system is observable (not detectable).
- Closed-Loop is stable by the separation principle (has Luenberger form).
- A Dual to the LQR problem. Replace (A,B,Q,R,K) with  $(A^T,C^T,V_1,V_2,L^T)$

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## Kalman Filter - Discrete Time

Assume the system is driven by noise  $w_k$  (no feedback)

$$x_{k+1} = Ax_k + w_k, \qquad y_k = Cx_k + v_k$$

The steady-state Kalman filter is an estimator of the form:

$$\hat{x}_{k+1} = A\hat{x}_k + L(C\hat{x}_k - y_k),$$

where  $v_k$  is sensor noise. This gives error  $(e_k = x_k - \hat{x}_k)$  dynamics

$$e_{k+1} = (A + LC)e_k$$

For the Kalman Filter, we choose  $L = A\Sigma C^T (C\Sigma C^T + V)^{-1}$  where  $V = \mathbf{E}[\mathbf{v_k v_k^T}]$  and  $\Sigma$  is the steady-state covariance of the error in the estimated state and satisfies

$$\Sigma = A\Sigma A^T + W - A\Sigma C^T (C\Sigma C^T + V)^{-1} C\Sigma A^T$$

where  $W = \mathbf{E}[\mathbf{w_k w_k^T}]$ . For the unsteady Kalman filter,  $\Sigma_k$  is updated at each time-step.

- If (A, W) controllable and (C, A) observable, then A + LC is stable.
- Again, dual to discrete-time LQR (which we haven't solved here!)